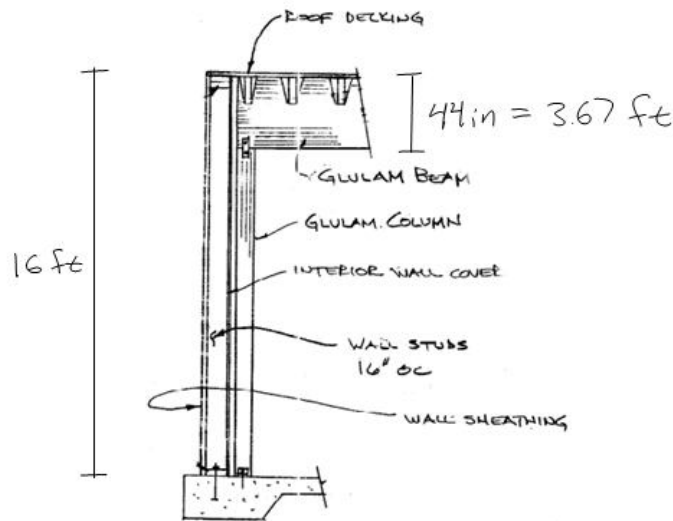
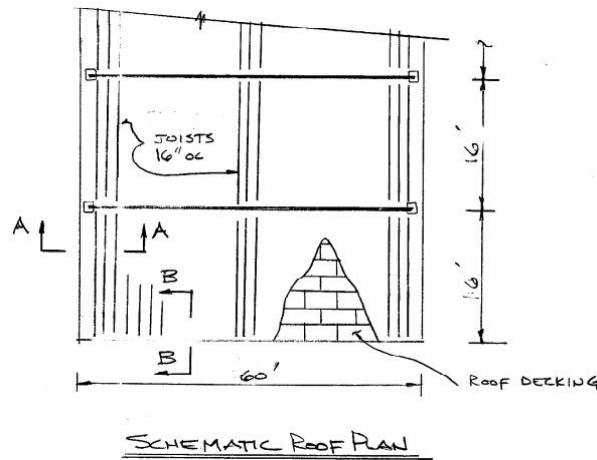


Given:



$$l = 16 - \frac{44}{12} = 12.333 \text{ ft}$$

$$t_w = 16 \text{ ft}$$

$$A_T = t_w \cdot 30 = 480 \text{ ft}^2$$

**Design Loads:** Use a 6.75" X 6.875" column for ease of connection with beam (11.5 plf).

· Dead Load: FBC 2005, Appendix A (P. A.1-A.3)

Plywood, rigid insulation, felt, gravel, HVAC, ceiling, joists, glulam beam (71 plf) and column self weight (11.5 plf)

$$D = 3 \cdot \frac{5}{8} + 4 \cdot .75 + 0.7 \cdot 5 + 104 \cdot \frac{3}{12} + 4 + 2 + 7 + \frac{71 \cdot 30}{16 \cdot 30} + \frac{11.5}{16} = 52.5 \text{ psf}$$

· Wind Load:

65.3 psf upward on 6 ft edge strips  
 55.1 psf upward on interior  
 19.4 psf downward

· Roof Live Load:

$$L_0 := 20 \text{ psf}$$

$$R_1 := \begin{cases} 1 & \text{if } A_T \leq 200 \\ (1.2 - 0.001 \cdot A_T) & \text{if } 200 < A_T < 600 \\ 0.6 & \text{if } A_T \geq 600 \end{cases} \quad R_1 = 0.72 \quad R_2 := 1.0$$

$$L_R := L_0 \cdot R_1 \cdot R_2 = 14.4 \text{ psf} \quad \text{ASCE 7-10, Eqn. (4.8-1)}$$

· Downward Load Cases:

*Dead Load Only:*

$$P_D := A_T \cdot D = 2.522 \times 10^4 \text{ lbs}$$

*Dead plus Roof Live:*

$$P_{D\_Lr} := A_T \cdot (D + L_R) = 3.213 \times 10^4 \text{ lbs}$$

*Dead plus 0.6\*Wind:*

$$P_{D\_W} := A_T \cdot (D + 0.6 \cdot 19.4) = 3.08 \times 10^4 \text{ lbs}$$

*Dead plus 3/4 of (Roof Live plus 0.6\*Wind):*

$$P_{D\_Lr\_W} := A_T \cdot [D + 0.75 \cdot (L_R + 0.6 \cdot 19.4)] = 3.459 \times 10^4 \text{ lbs}$$

· Upward Load Cases:                      0.6\*D+W                      [W-0.6\*D upward]                      C<sub>D</sub>=1.6

$$T := 16 \cdot 6 \cdot 65.3 + 16 \cdot 24 \cdot 55.1 - A_T \cdot 0.6 \cdot D = 1.23 \times 10^4 \text{ lbs}$$

· Estimate Controlling Load Case:

<u>Load Case:</u>	<u>Load (plf):</u>	<u>C<sub>D</sub>:</u>	<u>Load/C<sub>D</sub>:</u>
1. D	25220	0.9	28022
2. D + L <sub>R</sub>	32130	1.25	25704
5. D + 0.6*(W)	30800	1.6	19250
6a. D + 0.75*(L <sub>R</sub> + 0.6*W)	34590	1.6	21619

Therefore, for axial compression use:                      P := 25220 lbs                      C<sub>D</sub> := 0.9

**Check Axial Compression:**

**\*\*\*Use a 6.75" X 6.875" (5 laminations) Southern Pine Glulam Column in order to provide ease of connection with beam.**

NDS Supplement 2005, Table 1D, (P.25)

$$A : = 46.41 \text{ in}^2$$

$$r_x : = 1.985 \text{ in}$$

$$r_y : = 1.949 \text{ in}$$

$$d_x : = 6.875 \text{ in}$$

$$d_y : = 6.75 \text{ in}$$

· **Unbraced Length:**

$$l_x : = 1 \cdot 12 = 148 \text{ in}$$

$$l_y : = l_x = 148 \text{ in}$$

$$k : = 1.0 \quad \text{Assume pinned-pinned connection.}$$

NDS 2005, Appendix G (P.156)

· **Actual Compressive Stress:**

$$f_c : = \frac{P}{A} = 543.417 \text{ psi}$$

· **Allowable Compressive Stress:**

NDS 2005, Table 5.3.1 (P.34)

**\*\*\*Use Identification #47:**

$$F_c : = 1900 \text{ psi} \quad \text{4 or more laminations}$$

NDS Supplement 2005, Table 5B (P.66)

$$E_{\min} : = 730000 \text{ psi}$$

NDS Supplement 2005, Table 5B (P.66)

$$C_D : = 0.9$$

NDS 2005, Table 2.3.2 (P.9)

$$C_M : = 1.0 \quad \text{EMC} \leq 19\%$$

NDS Supplement 2005, Table 4B (P.37)

$$C_t : = 1.0$$

NDS 2005, Table 2.3.3 (P.9)

Column Stability Factor ( $C_P$ ):

NDS 2005, Section 3.7.1 (P.19)

$$l_{ex} := k \cdot l_x = 148 \text{ in} \qquad l_{ey} := k \cdot l_y = 148 \text{ in}$$

$$\frac{l_{ex}}{d_x} = 21.527$$

$$\frac{l_{ey}}{d_y} = 21.926 \quad \text{Controls}$$

$$F_{ce} := \frac{0.822 \cdot E_{min}}{\left(\frac{l_{ey}}{d_y}\right)^2} = 1.248 \times 10^3 \text{ psi}$$

$$[F_c'] := F_c \cdot C_D \cdot C_M \cdot C_t = 1.71 \times 10^3 \text{ psi}$$

$$c := 0.9 \quad \text{Structural Glulam}$$

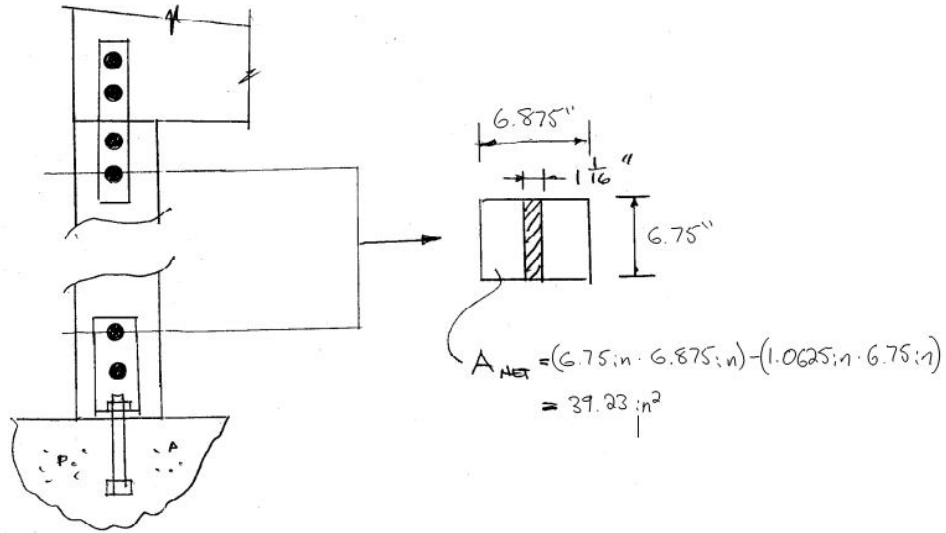
$$C_P := \frac{1 + \left(\frac{F_{ce}}{[F_c']}\right)}{2 \cdot c} - \sqrt{\left[\frac{1 + \left(\frac{F_{ce}}{[F_c']}\right)}{2 \cdot c}\right]^2 - \frac{\left(\frac{F_{ce}}{[F_c']}\right)}{c}} = 0.625$$

NDS 2005, Eqn. (3.7.1)

$$\text{Therefore:} \quad F'_c := F_c \cdot C_D \cdot C_M \cdot C_t \cdot C_P = 1.07 \times 10^3 \text{ psi} \quad > \quad f_c = 543.417 \text{ psi}$$

### Check Axial Tension

\*\*\*Assume a single row of 1"  $\phi$  bolts in 1-1/16"  $\phi$  holes.



$A_{net} : = 39.23 \text{ in}^2$

· Actual Tensile Stress:

$f_t : = \frac{T}{A_{net}} = 313.49 \text{ psi}$

· Allowable Tensile Stress:

NDS 2005, Table 5.3.1 (P.34)

$F_t : = 1200 \text{ psi}$

NDS Supplement 2005, Table 5B (P.66)

$C_D : = 1.6$

NDS 2005, Table 2.3.2 (P.9)

$C_M : = 1.0$

EMC  $\leq$  19%

NDS Supplement 2005, Table 4B (P.37)

$C_t : = 1.0$

NDS 2005, Table 2.3.3 (P.9)

Therefore:  $F'_t : = F_t \cdot C_D \cdot C_M \cdot C_t = 1.92 \times 10^3 \text{ psi} > f_t = 313.49 \text{ psi}$

### **Summary:**

- **Use a 6.75" X 6.875" (5 Laminations) Southern Pine Glulam Column.**
- **Use Identification #47.**